

CLEAN VERSION OF CLAIMS

1. A method for disposing of a rocket motor having a propellant contained therein and having an exhaust, comprising the steps of:

burning said propellant and concomitantly annularly spraying an enclosure of liquid completely surrounding the location in which the burning occurs.

2. A method according to claim 1, wherein the liquid includes at least one neutralising chemical for neutralising at least some noxious substances resulting from the burning or for capturing hazardous materials, or both.

3. A method according to claim 1, wherein said rocket contains a venturi mechanism, and wherein said venturi mechanism is removed prior to the burning step.

4. A method according to claim 3, wherein said rocket contains a venturi mechanism, and wherein said venturi mechanism is removed prior to the burning step.

5. A method according to claim 1, wherein the motor is secured in a substantially vertical position, with its exhaust end facing generally upwards, during the burning step.

6. A method according to claim 5, wherein the motor is clamped in a substantially vertical position, with its exhaust end facing generally upwards, during the burning step.

X21 A method according to claim *1*, wherein the motor is secured in a substantially vertical position, with its exhaust end facing generally upwards, during the burning step.

B4 *X22* A method according to claim 1, comprising further steps of filtering liquid from said enclosure and recycling the filtered liquid.

X23 *6* A method according to claim *1*, comprising further steps of filtering liquid from said enclosure and recycling the filtered liquid.

X24 The method according to claim 1 further characterized by deflecting the sprayed liquid to within a shroud or hood.

B5 *X25* *19* The method according to claim *1* further characterized by deflecting the sprayed liquid to within a shroud or hood.

X26 *20* The method according to claim *1* further characterized by deflecting the sprayed liquid to within a shroud or hood.